MTC OGMS/DINF

DIRECCION DE PERSONAL AERONAUTICO DPTO. DE INSTRUCCION PREGUNTAS Y OPCIONES POR TEMA

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TEMA: 0156	COMMERCIAL PILOT - (CH. 2) AIRCRAFT SYSTEMS	DDTA
	shutdown, while at idle, the ignition key is momentarily turned OFF. The engine continues to run with no tion; this	RPTA: B
-	rmal because the engine is usually stopped by moving the mixture to idle cut-off.	
OPCION B: show	ld not normally happen. Indicates a magneto not grounding in Off position	
OPCION C: is an	undesirable practice, but indicates that nothing is wrong.	
5170 Leaving	the carburetor heat on while taking off	C
	the mixture for more power on takeoff.	
	decrease the takeoff distance. Increase the ground roll.	
	o detect a broken magneto primary grounding lead is to he engine and momentarily turn the ignition off.	A
	full power, while holding the brakes, and momentarily turn off the ignition.	
	on one magneto, lean the mixture, and look for a rise in manifold pressure.	
5172 Fouling	of spark plugs is more apt to occur if the aircraft	A
	altitude with no mixture adjustment.	
	ends from altitude with no mixture adjustment.	
	tle is advanced very abruptly.	
	st probable reason an engine continues to run after the ignition switch has been turned off is	С
	on deposits glowing on the spark plugs. gneto ground wire is in contact with the engine casing.	
	ken magneto ground wire.	
	ound wire between the magneto and the ignition switch becomes disconnected, the engine	C
_	not operate on one magneto.	C
	ot be started with the switch in the BOTH position.	
OPCION C: coul	accidentally start if the propeller is moved with fuel in the cylinder.	
5175 For inte	rnal cooling, reciprocating aircraft engines are especially dependent on	В
	perly functioning cowl flap augmenter.	
	irculation of lubricating oil.	
	roper freon/compressor output ratio.	
•	t controls the air/fuel ratio with the	С
OPCION A: throt OPCION B: man	fold pressure.	
	ure control.	
5183 Which	tatement best describes the operating principle of a constant-speed propeller?	C
	arottle setting is changed by the pilot, the prop governor causes pitch angle of the propeller blades to	
	in unchanged.	
	gh blade angle, or increased pitch, reduces the propeller drag and allows more engine power for	
opcion c: take	oropeller control regulates the engine RPM and in turn the propeller RPM.	
	off equipped with constant-speed propellers and normally-aspirated engines, which procedure should be used	В
	placing undue stress on the engine components? When power is being	D
OPCION A: decr	eased, reduce the RPM before reducing the manifold pressure.	
	ased, increase the RPM before increasing the manifold pressure.	
	ased or decreased, the RPM should be adjusted before the manifold pressure.	
	Detonation may occur at high-power settings when	A
	uel mixture instantaneously ignites instead of burning progressively and evenly. cessively rich fuel mixture causes an explosive gain in power.	
OPCION B: an ex	accessivery from fuer miniture causes an expressive gain in power.	

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OPCION C: the fuel mixture is ignited too early by hot carbon deposits in the cylinder.	
5186 The uncontrolled firing of the fuel/air charge in advance of normal spark ignition is known as OPCION A: instantaneous combustion.	С
OPCION B: detonation. OPCION C: pre-ignition.	
5187 Fuel/air ratio is the ratio between the	В
OPCION A: volume of fuel and volume of air entering the cylinder.	
OPCION B: weight of fuel and weight of air entering the cylinder. OPCION C: weight of fuel and weight of air entering the carburetor.	
5188 The mixture control can be adjusted, which	A
OPCION A: prevents the fuel/air combination from becoming too rich at higher altitudes.	A
OPCION B: regulates the amount of air flow through the carburetor's venturi.	
OPCION C: prevents the fuel/air combination from becoming lean as the airplane climbs.	
5189 Which statement is true concerning the effect of the application of carburetor heat?	A
OPCION A: It enriches the fuel/air mixture.	
OPCION B: It leans the fuel/air mixture.	
OPCION C: It has no effect on the fuel/air mixture.	
5190 Detonation occurs in a reciprocating aircraft engine when	С
OPCION A: there is an explosive increase of fuel caused by too rich a fuel/air mixture. OPCION B: the spark plugs receive an electrical jolt caused by a short in the wiring.	
OPCION C: the unburned fuel/air charge in the cylinders is subjected to instantaneous combustion.	
5235 Propeller efficiency is the	A
OPCION A: ratio of thrust horsepower to brake horsepower.	
OPCION B: actual distance a propeller advances in one revolution.	
OPCION C: ratio of geometric pitch to effective pitch.	
5236 A fixed-pitch propeller is designed for best efficiency only at a given combination of	В
OPCION A: altitude and RPM.	
OPCION B: airspeed and RPM.	
OPCION C: airspeed and altitude.	
5237 The reason for variations in geometric pitch (twisting) along a propeller blade is that it	С
OPCION A: permits a relatively constant angle of incidence along its length when in cruising flight. OPCION B: prevents the portion of the blade near the hub from stalling during cruising flight.	
OPCION C: permits a relatively constant angle of attack along its length when in cruising flight.	
5271 A detuning of engine crankshaft counterweights is a source of overtsress that may be caused by	A
OPCION A: rapid opening and closing of the throttle.	
OPCION B: carburetor ice forming on the throttle valve.	
OPCION C: operating with an excessively rich fuel/air mixture.	
5298 The best power mixture is that fuel/air ratio at which	В
OPCION A: cylinder head temperatures are the coolest.	
OPCION B: the most power can be obtained for any given throttle setting.	
OPCION C: a given power can be obtained with the highest manifold pressure or throttle setting.	
5299 Detonation can be caused by	С
OPCION A: A rich mixture OPCION B: low engine temperatures.	
OPCION C: using a lower grade fuel than recommended.	
5606 Applying carburetor heat will	С
OPCION A: not affect the mixture.	C
OPCION B: lean the fuel/air mixture.	

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enrich the fuel/air mixture.	
an abnormally high engine oil temperature indication may be caused by	В
a defective bearing.	
operating with an excessively rich mixture.	
What will occur if no leaning is made with the mixture control as flight altitude increases?	C
The volume of air entering the carburetor decreases and the amount of fuel decreases.	
	C
· · · · · · · · · · · · · · · · · · ·	
	A
*	
	В
	A
	C
	C
increasing RPM by decreasing propeller blade angle.	
decreasing RPM by decreasing propeller blade angle.	
decreasing RPM by increasing propeller blade angle.	
o develop maximum power and thrust, a constant-speed propeller should be set to a blade angle that will produce a	В
large angle of attack and low RPM.	
small angle of attack and high RPM.	
large angle of attack and high RPM.	
or takeoff, the blade angle of a controllable-pitch propeller should be set at a	A
small angle of attack and high RPM.	
large angle of attack and low RPM.	
large angle of attack and high RPM.	
	C
Which is true regarding preheating an aircraft during cold weather operations?	A
vinch is true regarding preneating an anciart during cold weather operations?	A
The cobin groups well as the engine should be probested	
The cabin area as well as the engine should be preheated. The cabin area should not be preheated with portable heaters.	
	an abnormally high engine oil temperature indication may be caused by a defective bearing, the oil level being too low. operating with an excessively rich mixture. What will occur if no leaning is made with the mixture control as flight altitude increases? The volume of air entering the carburetor decreases and the amount of fuel decreases. The density of air entering the carburetor decreases and the amount of fuel decreases. The density of air entering the carburetor decreases and the amount of fuel decreases. The density of air entering the carburetor decreases and the amount of fuel emanises constant. Interest adjusted, the fuel/air mixture becomes richer with an increase in altitude because the amount of fuel decreases while the volume of air decreases. remains constant while the volume of air decreases. remains constant while the density of air decreases. The basic purpose of adjusting the fuel/air mixture control at altitude is to decrease the fuel flow to compensate for decreased air density. decrease the amount of fuel in the mixture to compensate for increased air density. decrease the amount of fuel in the mixture will cause the engine to overtheat. fouling of spark plugs. engine to overtheat. or exhaust gases leaking into the cockpit. a power loss due to back pressure in the exhaust system. a cold-running engine due to the heat withdrawn by the heater. or establish a climb after takeoff in an aircraft equipped with a constant-speed propeller, the output of the engine is seduced to climb power by decreasing propeller blade angle. decreasing RPM by decreasing propeller blade angle. or exablish a climb after takeoff in an aircraft equipped with a constant-speed propeller, the output of the engine is educed to climb power by decreasing propeller blade angle. or takeoff, the blade angle of a tack and high RPM. large angle of attack and high RPM. large ang