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TEMA: 0114 ATP - (CHAP. 03) AERODYNAMICS COD_PREG: PREGUNTA: 8324 When are inboard ailerons normally used?	RPTA:
OPCION A: Low-speed flight only.	
OPCION B: High-speed flight only.	
OPCION C: Low-speed and high-speed flight.	
When are outboard ailerons normally used?	A
OPCION A: Low-speed flight only.	
OPCION B: High-speed flight only.	
OPCION C: Low-speed and high-speed flight.	
Which of the following is considered a primary flight control?	В
OPCION A: Slats.	
OPCION B: Elevator. OPCION C: Dorsal fin.	
8327 Which of the following is considered an auxiliary flight control?	С
OPCION A: Ruddervator. OPCION B: Upper rudder.	
OPCION C: Leading-edge flaps.	
8328 What is the purpose of a control tab?	A
OPCION A: Move the flight controls in the event of manual reversion.	А
OPCION B: Reduce control forces by deflecting in the proper direction to move a primary flight control.	
OPCION C: Prevent a control surface from moving to a full-deflection position due to aerodynamic forces.	
8329 What is the purpose of an anti-servo tab?	С
OPCION A: Move the flight controls in the event of manual reversion.	
OPCION B: Reduce control forces by deflecting in the proper direction to move a primary flight control.	
OPCION C: Prevent a control surface from moving to a full-deflection position due to aerodynamic forces.	
8330 What is the purpose of a servo tab?	В
OPCION A: Move the flight controls in the event of manual reversion.	
OPCION B: Reduce control forces by deflecting in the proper direction to move a primary flight control.	
OPCION C: Prevent a control surface from moving to a full-deflection position due to aerodynamic forces.	
Which is a purpose of leading-edge flaps?	A
OPCION A: Increase the camber of the wing.	
OPCION B: Reduce lift without increasing airspeed. OPCION C: Direct airflow over the top of the wing at high angles of attack.	
8332 What is a purpose of flight spoilers?	В
OPCION A: Increase the camber of the wing. OPCION B: Reduce lift without increasing airspeed.	
OPCION C: Direct airflow over the top of the wing at high angles of attack.	
8333 For which purpose may flight spoilers be used?	A
OPCION A: Reduce the wings' lift upon landing.	A
OPCION B: Increase the rate of descent without increasing aerodynamic drag.	
OPCION C: Aid in longitudinal balance when rolling an airplane into a turn.	
8334 Which is a purpose of leading-edge slats on high-performance wings?	C
OPCION A: Decrease lift at relative slow speeds.	2
OPCION B: Improve aileron control during low angles of attack.	
OPCION C: Direct air from the high pressure area under the leading edge along the top of the wing.	
8335 Which is a purpose of leading-edge slats on high-performance wings?	С
OPCION A: Decrease lift at relative slow speeds.	

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OPCION B: OPCION C:	Improve aileron control during low angles of attack. Direct air from the high pressure area under the leading edge along the top of the wing.	
8336 W	hich is a purpose of ground spoilers?	A
OPCION A:	Reduce the wings' lift upon landing.	
OPCION B:	Aid in rolling an airplane into a turn.	
OPCION C:	Increase the rate of descent without gaining airspeed.	
8337 W	hich direction from the primary control surface does an anti-servo tab move?	A
OPCION A:	Same direction.	
OPCION B:	Opposite direction.	
OPCION C:	Remains fixed for all positions.	
	hich direction from the primary control surface does a servo tab move?	В
OPCION A:	Same direction.	
OPCION B:	Opposite direction.	
OPCION C:	Remains fixed for all positions.	
	hich direction from the primary control surface does an elevator adjustable trim tab move when the control surface	C
OPCION A:	moved? Same direction.	
OPCION A:	Opposite direction.	
OPCION C:	Remains fixed for all positions.	
8340 W	hat is the purpose of an elevator trim tab?	(
OPCION A:	Provide horizontal balance as airspeed is increased to allow hands-off flight.	
OPCION B:	Adjust the speed tail load for different airspeeds in flight allowing neutral control forces.	
OPCION C:	Modify the downward tail load for various airspeeds in flight eliminating flight-control pressures.	
8341 W	hich is a purpose of wing-mounted vortex generators?	A
OPCION A:	Reduce the drag caused by supersonic flow over portions of the wing.	
OPCION B:	Increase the onset of drag divergence and aid in aileron effectiveness at high speed.	
OPCION C:	Break the airflow over the wing so the stall will progress from the root out to the tip of the wing.	
8342 W	hy do some airplanes equipped with inboard/outboard ailerons use the outboards for slow flight only?	Е
OPCION A:	Increased surface area provides greater controllability with flap extension.	
OPCION B:	Aerodynamic loads on the outboard ailerons tend to twist the wingtips at high speeds.	
OPCION C:	Locking out the outboard ailerons in high-speed flight provides variable flight control feel.	
	hich of the following are considered primary flight controls?	C
OPCION A:	Tabs.	
OPCION B:	Flaps.	
OPCION C:	Outboard ailerons.	
	hat effect does an increase in airspeed have on a coordinated turn while maintaining a constant angle of bank and citude?	C
OPCION A:	The rate of turn will decrease resulting in a decreased load factor.	
OPCION B:	The rate of turn will increase resulting in a increased load factor.	
OPCION C:	The rate of turn will decrease resulting in no changes in load factor.	
	hat is the effect on total drag of an aircraft if the airspeed decreases in level flight below that speed for maximum D?	A
OPCION A:	Drag increases because of increased induced drag.	
OPCION B:	Drag increases because of increased parasite drag.	
OPCION C:	Drag decreases because of lower induced drag.	
8347 W	hat is load factor?	(
OPCION A:	Lift multiplied by the total weight.	
OPCION B:	Lift substracted from the total weight.	
OPCION C:	Lift divided by the total weight.	

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8348 What affects indicated stall speed?	A
OPCION A: Weight, load factor, and power.	7.1
OPCION B: Load factor, angle of attack, and power.	
OPCION C: Angle of attack, weight, and air density.	
8349 If no corrective action is taken by the pilot as angle of bank is increased, how is the vertical component of lift and sink rate affected?	С
OPCION A: Lift increases and the sink rate increases.	
OPCION B: Lift decreases and the sink rate decreases.	
OPCION C: Lift decreases and the sink rate increases.	
8350 Why must the angle of attack be increased during a turn to maintain altitude?	A
OPCION A: Compensate for loss of vertical component of lift.	
OPCION B: Increase the horizontal component of lift equal to the vertical component.	
OPCION C: Compensate for increase in drag.	
8351 How can the pilot increase the rate of turn and decrease the radius at the same time?	В
OPCION A: Steepen the bank and increase airspeed.	
OPCION B: Steepen the bank and decrease airspeed. OPCION C: Shallow the bank and increase airspeed.	
•	
What is the relationship of the rate of turn with the radius of turn with a constant angle of bank but increasing airspeed?	A
OPCION A: Rate will decrease and radius will increase.	
OPCION B: Rate will increase and radius will decrease.	
OPCION C: Rate and radius will increase.	
8353 Upon which factor does wing loading during a level coordinated turn in smooth air depend?	В
OPCION A: Rate of turn.	
OPCION B: Angle of bank.	
OPCION C: True airspeed.	
8354 If an aircraft with a gross weight of 2,000 pounds were subjected to a total load of 6,000 pounds in flight, the load factor would be	В
OPCION A: 2 Gs.	
OPCION B: 3 Gs.	
OPCION C: 9 Gs.	
8356 Airflow separation over the wing can be delayed by using vortex generators	C
OPCION A: directing high pressure air over the top of the wing or flap through slots and making the wing surface smooth.	
OPCION B: directing a suction over the top of the wing or flap through slots and making the wing surface smooth.	
OPCION C: making the wing surface rough and/or directing high pressure air over the top of the wing or flap through	
slots.	
8357 In a light, twin-engine airplane with one engine inoperative, when is it acceptable to allow the ball of a slip-skid	В
indicator to be deflected outside the reference lines?	
OPCION A: While maneuvering at minimum controllable airspeed to avoid overbanking. OPCION B: When operating at any airspeed greater than Vmc.	
OPCION C: When practicing immiment stalls in a banked attitude.	
8358 What is the safest and most efficient takeoff and initial climb procedure in a light, twin-engine airplane? Accelerate	С
to	
OPCION A: best engine-out, rate-of-climb airspeed while on the ground, then lift off and climb at that speed.	
OPCION 6: Vmc, then lift off at that speed and climb at maximum angle-of-climb airspeed.	
OPCION C: an airspeed slightly above Vmc, then lift off and climb at the best rate-of-climb airspeed.	
8359 What procedure is recommended for an engine-out approach and landing?	A
OPCION A: The flightpath and procedures should be almost identical to a normal approach and landing.	

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8360 What performance should a pilot of a light, twin-engine airplane be able to maintain at Vmc?	
OPCION A: Heading.	1
OPCION B: Heading and altitude.	
OPCION C: Heading, altitude, and ability to climb 50 ft/min.	
8361 What criteria determines which engine is the "critical" engine of a twin-engine airplane?	A
OPCION A: The one with the center of thrust closest to the centerline of the fuselage.	
OPCION B: The one designated by the manufacturer which develops most usable thrust.	
OPCION C: The one with the center of thrust farthest from the centerline of the fuselage.	
What effect, if any, does altitude have on Vmc for an airplane with unsupercharged engines?	(
OPCION A: None.	
OPCION B: Increases with altitude.	
OPCION C: Decreases with altitude.	
Under what condition should stalls never be practiced in a twin-engine airplane?	A
DPCION A: With one engine inoperative.	
OPCION B: With climb power on.	
OPCION C: With full flaps and gear extended.	
What does the blue radial line on the airspeed indicator of a light, twin-engine airplane represent?	1
OPCION A: Maximum single-engine rate of climb.	
DPCION B: Maximum single-engine angle of climb. DPCION C: Minimum controllable airspeed for single-engine operation.	
8365 Identify the type stability if the aircraft attitude remains in the new position after the controls have been neutralized.	(
OPCION A: Negative longitudinal static stability.	
OPCION B: Neutral longitudinal dynamic stability. OPCION C: Neutral longitudinal static stability.	
8366 What is a characteristic of longitudinal instability? OPCION A: Pitch oscillations becoming progressively greater.	I
OPCION B: Bank oscillations becoming progressively greater.	
OPCION C: Aircraft constantly tries to pitch down.	
8367 Describe dynamic longitudinal stability.]
OPCION A: Motion about the longitudinal axis.	•
OPCION B: Motion about the lateral axis.	
OPCION C: Motion about the vertical axis.	
8368 What is the reason for variations in geometric pitch along a propeller or rotor blade?	
OPCION A: It permits a relatively constant angle of attack along its length when in cruising flight.	
OPCION B: It prevents the portion of the blade near the hub or root from stalling during cruising flight.	
OPCION C: It permits a relatively constant angle of incidence along its length when in cruising flight.	
8372 Identify the type stability if the aircraft attitude tends to move farther from its original position after the controls	I
have been neutralized.	
OPCION A: Negative static stability.	
OPCION B: Positive static stability.	
OPCION C: Negative dynamic stability.	
8373 Identify the type stability if the aircraft attitude tends to return to its original position after the controls have been neutralized.]
OPCION A: Positive dynamic stability.	
OPCION B: Positive static stability.	
OPCION C: Neutral dynamic stability.	

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8375 What flight condition should be expected when an aircraft leaves ground effect? A **OPCION A:** An increase in induced drag requiring a higher angle of attack. **OPCION B:** A decrease in parasite drag permitting a lower angle of attack. An increase in dynamic stability. **OPCION C:** 8376 What characteristic should exist if an airplane is loaded to the rear of its CG range? C **OPCION A:** Sluggish in aileron control. **OPCION B:** Sluggish in rudder control. **OPCION C:** Unstable about the lateral axis. 8377 What will be the ratio between airspeed and lift if the angle of attack and other factors remain constant and airspeed C is doubled? Lift will be OPCION A: the same. **OPCION B:** two times greater. **OPCION C:** four times greater. 8378 What true airspeed and angle of attack should be used to generate the same amount of lift as altitude is increased? В OPCION A: The same true airspeed and angle of attack. **OPCION B:** A higher true airspeed for any given angle of attack. **OPCION C:** A lower true airspeed and higher angle of attack. 8379 How can an airplane produce the same lift in ground effect as when out of ground effect? R The same angle of attack. OPCION A: **OPCION B:** A lower angle of attack. A higher angle of attack. **OPCION C:** 8380 What are some characteristics of an airplane loaded with the CG at the aft limit? A Lowest stall speed, highest cruise speed, and least stability. **OPCION B:** Highest stall speed, highest cruise speed, and least stability. **OPCION C:** Lowest stall speed, lowest cruise speed, and highest stability. 8382 By changing the angle of attack of a wing, the pilot can control the airplane's R OPCION A: lift, gross weight, and drag. **OPCION B:** lift, airspeed, and drag. **OPCION C:** lift and airspeed, but not drag. В 8384 The primary purpose of high-lift devices is to increase the OPCION A: L/Dmax. **OPCION B:** lift at low speeds. **OPCION C:** drag and reduce airspeed. 8385 What is the primary function of the leading edge flaps in landing configuration during the flare before touchdown? OPCION A: Prevent flow separation. **OPCION B:** Decrease rate of sink. **OPCION C:** Increase profile drag. 8386 What effect does the leading edge slot in the wing have on performance? В OPCION A: Decreases profile drag. **OPCION B:** Changes the stalling angle of attack to a higher angle. Decelerates the upper surface boundary layer air. **OPCION C:** 8387 Within what Mach range does transonic flight regimes usually occur? В OPCION A: .50 to .75 Mach. **OPCION B:** .75 to 1.20 Mach. **OPCION C:** 1.20 to 2.50 Mach. 8388 What is the highest speed possible without supersonic flow over the wing? В Initial buffet speed. OPCION A: **OPCION B:** Critical Mach number.

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OPCION C: Transonic index.	
8389 What is the free stream Mach number which produces first evidence of local sonic flow?	С
OPCION A: Supersonic Mach number.	
OPCION B: Transonic Mach number.	
OPCION C: Critical Mach number.	
8390 At what Mach range does the subsonic flight range normally occur?	A
OPCION A: Below .75 Mach.	
OPCION B: From .75 to 1.20 Mach.	
OPCION C: From 1.20 to 2.50 Mach.	
What is the principal advantage of a sweepback design wing over a straightwing design?	A
OPCION A: The critical Mach number will increase significantly.	
OPCION B: Sweepback will increase changes in the magnitude of force coefficients due to compressibility.	
OPCION C: Sweepback will accelerate the onset of compressibility effect.	
What is the result of a shock-induced separation of airflow occurring symmetrically near the wing root of a	В
sweptwing aircraft?	
OPCION A: A high-speed stall and sudden pitchup. OPCION B: A severe moment or "tuck under".	
OPCION 6: A severe moment of tuck under : OPCION C: Severe porpoising.	
8393 What is one disadvantage of a sweptwing design?	В
OPCION A: The wing root stalls prior to the wingtip section.	Б
OPCION A: The wing root stans prior to the wingup section. OPCION B: The wingtip section stalls prior to the wing root.	
OPCION C: Severe pitchdown moment when the center of pressure shifts forward.	
· · · · · · · · · · · · · · · · · · ·	
8394 What is the condition known as when gusts cause a sweptwing-type airplane to roll in one direction while yawing in the other?	С
OPCION A: Porpoise.	
OPCION B: Wingover.	
OPCION C: Dutch roll.	
8395 What is the movement of the center of pressure when the wingtips of a sweptwing airplane are shock-stalled first?	В
OPCION A: Inward and aft.	
OPCION B: Inward and forward.	
OPCION C: Outward and forward.	
8396 For a given angle of bank, the load factor imposed on both the aircraft and pilot in a coordinated constant-altitude	С
turn	
OPCION A: is directly related to the airplane's gross weight.	
OPCION B: varies with the rate of turn.	
OPCION C: is constant.	
What is the relationship between induced and parasite drag when the gross weight is increased?	В
OPCION A: Parasite drag increases more than induced drag.	
OPCION B: Induced drag increases more than parasite drag.	
OPCION C: Both parasite and induced drag are equally increased.	
8399 At which speed will increasing the pitch attitude cause an airplane to climb?	В
OPCION A: Low speed.	
OPCION B: High speed.	
OPCION C: Any speed.	